NASA TECHNICAL NOTE



NASA TN D-6272

H LIBRARY KAFB, NM

LOAN COPY: RETURN T AFWL (DOGL) KIRTLAND AFB, N. M.

COMPILATION OF XENON FLASH TUBE DATA FOR PILOT WARNING INDICATOR SYSTEMS

by Robert Gagnon and Anthony R. Lewis Electronics Research Center Cambridge, Mass. 02139



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION . WASHINGTON, D. C. . MARCH 1971

				07330					
1. Report No. NASA TN D-6272	2. Government Acc	cession No.	3. Recipient's Catalo	og No.					
4. Title and Subtitle Compilation of Xer	on Flash T		5. Report Date March 1971						
for Pilot Warning		1 4	6. Performing Organization Code						
7. Author(s) Robert Gagnon and	Anthony R.		3. Performing Organi	zation Report No.					
9. Performing Organization Name and	Address	10	D. Work Unit No.	01 05					
Electronics Resear	1	125-22-07- 1. Contract or Grant							
Cambridge, Massach	usetts	1:	13. Type of Report and Period Cov						
12. Sponsoring Agency Name and Addr	ess		m1 2 1	37 - 1 -					
National Aeronauti		[1	Technical						
Space Administrati Washington, D. C. 209			14. Sponsoring Agency Code						
15. Supplementary Notes									
as aircraft beacor (2 units) which was lamp output was me 580-23A), factory-Four filters (cent and 0.974 µm) were characteristics. were: pulse shape steradiancy, peak peak optical power output, and conver	as not desing as not desing as a second with the contract of t	gned for the day and a radiomed against a radiomed	nat purpose eter (EG&G secondary 5, 0.876, (detailed s d for each rise time lant intens	e. Flash Model standard 0.927, spectral lamp , lamp sity,					
17. Key Words Xenon Flash Tube PWI Systems		18. Distribution State	ement						
		Unclas	sified - Unl	imited					
19. Security Classif, (of this report)	20. Security Clas		sified - Unl	imited					
19. Security Classif. (of this report) Unclassified									

^{*}For sale by the National Technical Information Service, Springfield, Virginia 22151

COMPILATION OF XENON FLASH TUBE DATA FOR PILOT WARNING INDICATOR SYSTEMS

by Robert Gagnon and Anthony R. Lewis

SUMMARY

To provide design data for Pilot Warning Indicator (PWI) systems, we measured the characteristics of six types of xenon strobe lamps (14 units) built to operate as aircraft beacons and one type of xenon strobe lamp (2 units) which was not designed for that purpose. Flash lamp output was measured with a radiometer (EG&G model 580-23A), factory-calibrated against a secondary standard. Four filters (central wavelengths 0.825, 0.876, 0.927, and $0.974~\mu m$) were used to obtain more detailed spectral characteristics. Properties determined for each lamp were: pulse shape, pulse width, pulse rise time, lamp steradiancy, peak irradiance, peak radiant intensity, peak optical power, average optical power, optical energy output, and conversion efficiency.

INTRODUCTION

A Pilot Warning Indicator (PWI) system, using xenon flash tubes as radiation sources, is presently being developed by NASA. To facilitate the design of detectors for the system, we have measured the characteristics of commercially available xenon flash tubes, primarily those currently in use as aircraft beacons. Of particular interest was the measurement of radiated power to set acceptance thresholds and the determination of pulse-width and pulse rise-time for pulse-discrimination circuits which improve the signal-to-noise ratio.

The information presented here is for specific flash tubes; manufacturers may change specifications from time to time and variation in production runs is to be expected. Our data are intended as an aid to designers and should, in no way, be regarded as rating one tube against another.

EXPERIMENTS

A diagram of the experimental setup is shown in Figure 1. The optical bench ensured the stability and constant separation distance of the radiometer and xenon lamp system. The radiometer (EG&G model 580-23A, S/N385) was factory-calibrated against a secondary standard, Figure 2. The oscilloscope (Tektronix 502A,

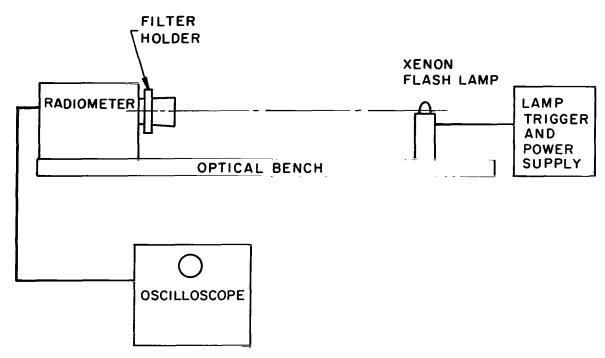
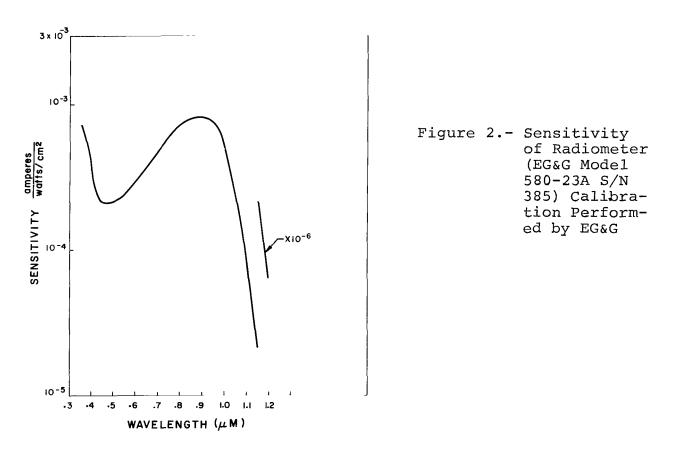


Figure 1.- Experimental Set Up



Dual-beam S/N 025935) was calibrated at NASA, ERC, just prior to the tests. Pulses were recorded with an oscilloscope camera for analysis. Filters could be inserted in the receptor cone of the radiometer to observe the output in a specific passband. Four filters were used with central wavelengths of 0.825, 0.876, 0.927, and 0.974 $\mu m;$ the filter transmission characteristics are shown in Figure 3.

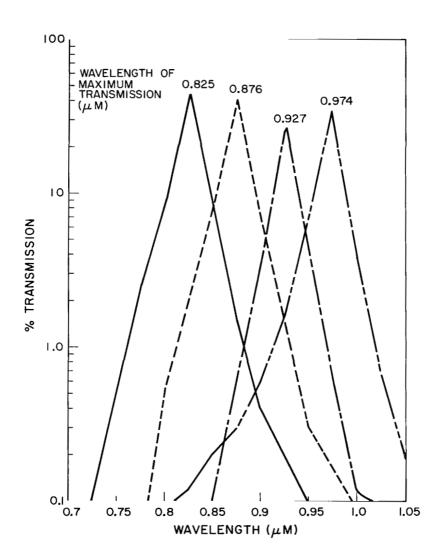


Figure 3.- Optical Filter Transmission Characteristics (Filters from Optical Technology, Inc.)

Measurement

Pulse form.— The pulse shapes of the total unfiltered radiated power are shown in Figures 4 through 10. The computed rise-times and pulse widths are given in Table I. Rise-time is taken as the time interval between the 10 percent and 90 percent amplitude points on the leading edge of the pulse. The pulse width, τ , is defined as:

$$\tau = \frac{\left| \int_{\tau_0}^{\tau_1} V(t) dt \right|^2}{\int_{\tau_0}^{\tau_1} |V(t)|^2 dt}$$
(1)

where V(t) is the voltage developed across the sampling resistor at the radiometer output and τ_1 - τ_0 is the total width of the pulse given by the baseline.

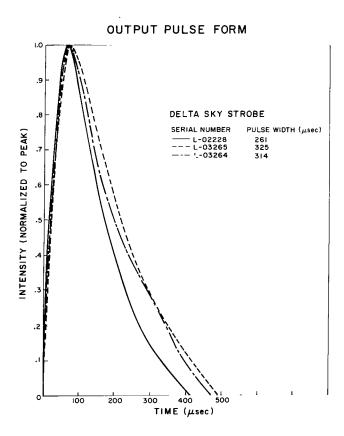


Figure 4.- Output Pulse Form

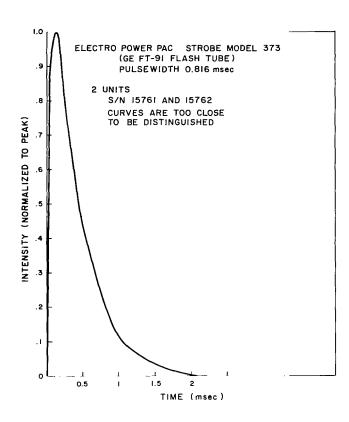
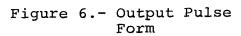
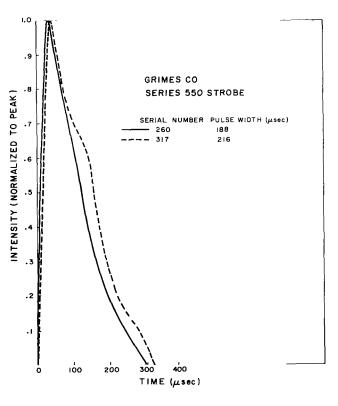


Figure 5.- Output Pulse Form





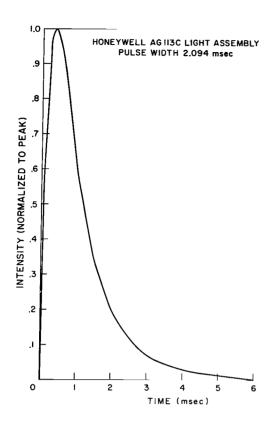
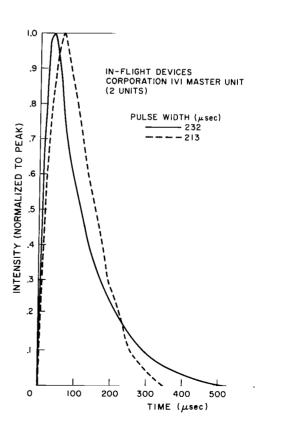


Figure 7.- Output Pulse Form

Figure 8.- Output Pulse Form



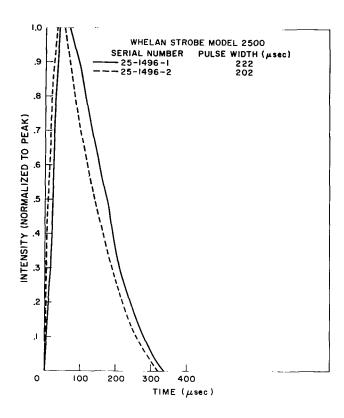
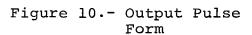


Figure 9.- Output Pulse Form



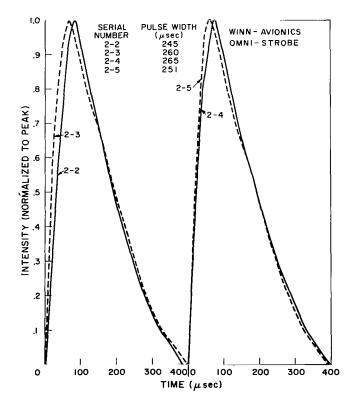


Table I.- Pulse Form Characteristics of Some Xenon Flash Tubes

Manufacturer	Model	Serial Number	Telec (Joules)	Rise Time (µsec)	Pulse Width (µsec)	Notes
Delta	Skystrobe	L-02228 L-03264 L-03265	17.8 18.4 18.4	44 43 52	261 314 325	160µF Capa- citor
Electro Power Pac	373	15761 15761 15762	5.47 17.3 5.47	40 40 40	816 816 816	with GE FT-91 Flash Tube
Grimes Co.	Series 550	260 317	20.7 20.7	22 31	188 216	
Honeywell	Wingtip AG113C Light Assembly	7792	14.1	360	2094	
In-Flight Devices	IVI Master Unit	16JE 6-3	10.95 10.95	21 45	232 213	
Whelan	2500	25-1496-1 25-1496-2	28 28	26 25	222 202	
Winn Avionics	Omnistrobe	2-2 2-3 2-4 2-5	10.6 10.6 10.6 10.6	54 39.5 47.5 37.5	245 260 265 251	

- 1. All units designed for aircraft except Electro-Power Pac.
- 2. Rise time taken between 10 percent and 90 percent amplitude points on leading edge of pulse
- 3. Pulse width defined in text Eq. (1).

The value of pulse-width, as defined in Eq. (1), is relatively insensitive to transient fluctuations near the peak whereas a definition of pulse width based upon the width at some fraction, say one-half of peak width is highly sensitive to just what value is selected for the maximum. Also, the pulse width as defined in Eq. (1) is approximately equal (for fast rise-times) to twice the RC time-constant of the lamp system.

The integrals were evaluated graphically using a compensating polar planimeter. With the observed xenon lamp pulse shapes this pulse-width generally corresponds to about the 20 percent amplitude level.

The following spectral characteristics were computed from the experimental data: peak irradiance, radiant intensity, peak optical lamp power, average optical lamp power, optical energy output, and conversion efficiency. These values are summarized in Table II.

Peak irradiance. The peak energy flux at the detector, in the passband of the filter.

$$E_{\Delta\lambda} = \frac{V_{P}}{R_{L} S_{\Delta\lambda}}$$
 (2)

where: $E_{\Delta\lambda}$ is the peak irradiance at the detector, watts/cm² $V_{\rm p}$ is the pulse maximum (from oscilloscope), volts $R_{\rm L}$ is the load resistor at the radiometer output, ohms $S_{\Delta\lambda}$ is the average detector sensitivity in the passband, amps/watt/cm² (fig. 2).

Peak Radiant Intensity. The peak power output of the lamp radiated into a unit solid angle in the filter passband.

$$I_{\Delta\lambda} = \frac{E_{\Delta\lambda} d^2}{T_{\text{eff}\Delta\lambda}}$$
 (3)

$$T_{\text{eff}\Delta\lambda} = \frac{\int_{\lambda_{1}}^{\lambda_{2}} T(\lambda)P(\lambda)d\lambda}{\int_{\lambda_{1}}^{\lambda_{2}} P(\lambda)d\lambda}$$
(4)

where $T(\lambda)$ is the filter transmission as a function of wavelength, Figure 3

 $P(\lambda)$ is the xenon lamp optical output power spectrum, Figure 11

 λ_1 , λ_2 are the 3 percent points of the product curve $T(\lambda)P(\lambda)$, Figure 11

The unfiltered radiated output power spectrum was measured for a typical lamp (GE FT-91 flash tube used with the Electro-Power Pac system); it was measured against a calibrated lamp through a Jarrell-Ash monochromator system. The lamp was operated at current densities comparable with those of aircraft strobe lights; intensities were taken at 10Å (l μm) intervals. The values obtained for each wavelength interval was then multiplied by the filter transmission characteristic and the integrals in Eq. (4) evaluated graphically in the manner previously described.

Peak Optical Lamp Power. The peak power output at the lamp in the filter passband.

$$P_{\Delta\lambda} = T_{\Delta\lambda} \Omega_{L}$$
 (5)

where

 ${\rm P}_{\Delta\lambda}$ is the peak optical lamp power, watts ${\rm \Omega}_{\rm L}$ is the solid angle into which the lamp radiates, steradians

The steradiancy of the lamps were calculated geometrically from the restricting mechanical apertures. All measurements and calculations were made for the radiometer and lamp in the same horizontal plane, with their axes oriented on the line connecting the center of the lamp with the center of the radiometer's detecting surface. No effort was made to locate a maximum energy point by rotating or tilting the xenon lamp.

Optical Energy Output. The total energy radiated by the lamp in the filter passband is given by:

$$J_{op\Delta\lambda} = P_{\Delta\lambda} \int_{1 \text{ cycle}} U(t) dt$$
 (6)

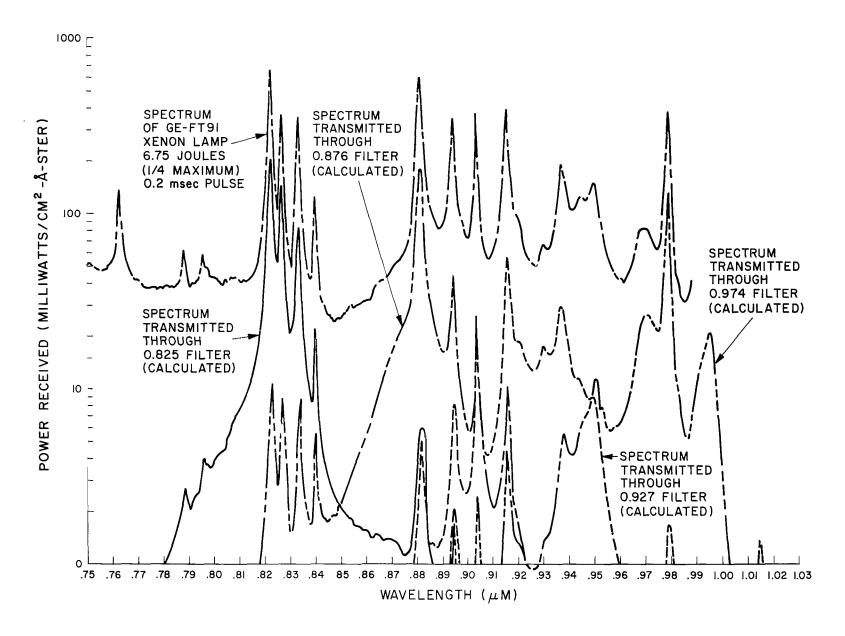


Figure 11.- Xenon Lamp Spectrum-Unfiltered and Transmitted Through Filters (Central Wavelengths 0.825, 0.876, 0.927 and 0.974 µMeters)

where $J_{\text{op}\Delta\lambda}$ is the optical energy output, joules U(t) is the observed pulse form normalized to unify at the peak

Average Optical Power. - Average optical power is shown as:

$$P_{ave\Delta\lambda} = \frac{J_{op\Delta\lambda}}{\tau}$$
 (7)

Conversion Efficiency. The efficiency with which the lamp system converts the stored electrical energy into radiated energy in the filter passband

$$\varepsilon_{\Delta\lambda} = 100 \frac{J_{\text{op}\Delta\lambda}}{J_{\text{elec}}}$$
(8)

where $\epsilon_{\Delta\lambda}$ is the conversion efficiency, percent (%) $\rm J_{elec}$ is the stored energy, joules

The stored electrical energy is given by

$$J_{\text{elec}} = \frac{1}{2} CV^2$$
 (9)

where C is the capacitance connected directly across the lamp, farads

V is the voltage across the capacitor terminals just prior to discharge, volts

TABLE II. SPECTRAL CHARACTERISTICS OF SOME XENON FLASH TUBES (Continued)

		Serial		Peak Optical Power (watts) $^{ m P}_{\Delta\lambda}$				Average Optical Power (watts) ^P aveΔλ				Optical Energy (Joules) ^J ορΔλ				Conversion Efficiency (%) $^{arepsilon} \lambda$			
Manufacturer	Model	Number	Notes	0.825	0.876	0.927	0.974	0.825	0.876	0.927	0.974	0.825	0.876	0.927	0.974	0.825	0.876	0.927	0.974
Delta	Skystrobe	L-02228 L-02228 L-03264 L-03265	1 2	1015 921 870 790	1208 1203 1201 1112	1985 1843 1680 1515	2193 1898 1700 1594	664 602 564 524	770 786 778 738	1297 1205 1089 1006	1433 1241 1102 1058	0.17 0.16 0.18 0.17	0.21 0.21 0.24 0.24	0.34 0.31 0.34 0.33	0.37 0.32 0.35 0.34	0.96 0.90 0.96 0.93	1.2 1.2 1.33 1.3	1.9 1.8 1.86 1.78	2.1 1.8 1.88 1.89
Electro Power Pac	373	15761 15761 15762 15762 15762	3,4,7 3,5,7 3,4,7 3,4,6,7 3,5,6,7	688 2447 890 109 376	951 3609 1254 152 524	1407 5391 1886 223 767	1457 5962 2012 238 729	421 1540 560 68.8 237	598 2271 789 95.7 330	886 3392 1187 141 483	917 3752 1266 150 459	0.34 1.26 0.46 0.056 0.19	0.49 1.85 0.64 0.078 0.27	0.72 2.77 0.97 0.115 0.39	0.75 3.06 1.03 0.122 0.37	6.3 7.3 8.4 1.0 1.1	8.9 10.7 11.8 1.4 1.6	13.2 16.0 17.7 2.1 2.3	13.7 17.7 18.9 2.2 2.2
Grimes Co.	Series 550	260 317		3584 3177	4731 4355	6849 6316	7207 6983	2362 2063	3118 2827	4514 4100	4749 4533	0.44	0.59 0.61	0.85 0.88	0.89 0.98	2.15 2.15	2.83 2.95	3.68 4.28	3.83 4.73
Honeywell	Wingtip AG 113C Light Assembly	7792		21.0	30.4	46.8	51.9	13.4	19.3	29.7	33.0	0.028	0.040	0.062	0.069	0.2	0.29	0.44	0.49
In-Flight Devices	IVI Master Unit	16JE6-3	2	1172 1175 1247	1597 1621 1713	2408 2370 2574	2419 2440 2418	683 686 851	932 945 1170	1404 1382 1759	1410 1423 1651	0.16 0.16 0.18	0.22 0.22 0.25	0.33 0.32 0.37	0.33 0.33 0.35	1.45 1.45 1.66	1.97 2.00 2.28	2.98 2.93 3.42	2.99 3.02 3.21
Whelan	2500	25-1496-1 25-1496-2		3516 3191	4702 4324	6429 6211	6643 6411	2542 2205	3400 2988	4648 4292	4802 4430	0.56 0.45	0.76 0.60	1.03 0.87	1.07 0.90	2.01 1.61	2.7 2.14	3.68 3.11	3.83 3.22
Winn Co.	Omnistrobe	2-2 2-3 2-4 2-5		1164 943 876. a 859.1	1640 1281 1167 1218	2330 1780 1587 1644	2440 1915 1797 1797	779.3 649.6 579.8 591.8	1097 882 772.1 838.7	1560 1225 1050 1133	1633 1318 1189 1238	0.191 0.169 0.154 0.149	0.269 0.229 0.205 0.211	0.382 0.319 0.278 1.284	0.400 0.343 0.315 0.311	1.8 1.59 1.45 1.40	2.54 2.16 1.93 1.99	3.61 3.01 2.62 2.68	3.77 3.23 2.97 2.93

Notes 1 Special 160 uF Capacitor
2 Viewed at 90° with respect to
preceeding measurement
3 With GE FT-91 Flash Tube
4 Jelec = 5.47 Joules

5 J_{elec} = 17.3 Joules

6 No Reflector
7 Not Designed for Aircraft

 $C_{\mathbf{I}}$.125

TABLE II. SPECTRAL CHARACTERISTICS OF SOME XENON FLASH TUBES

Serial				Lamp to Detector Separation (cm)	r Lamp on Steradiancy (Steradians)	Pulse Peak from Oscilloscope (mvolts) V _p				Peak Irradiance at Detector (mwatts/cm ²) Ε _{Δλ}				Peak Radiant Intensity (watts/ster) ${}^{ m I}_{\Delta\lambda}$				
Manufacturer	Model	Number	Notes	đ	$\Omega^{\mathbf{L}}$	0.825	0.876	0.927	0.974	0.825	0.876	0.927	0.974	0.825	0.876	0.927	0.974	
Delta	Skystrobe	L-02228 L-02228 L-03264 L-03265	1 2	126.8 127.0 139.1 138.4	9.55 9.55 9.55 9.55	1.04 0.94 0.74 0.68	0.86 0.85 0.71 0.66	0.64 0.59 0.45 0.41	0.64 0.55 0.41 0.39	2.67 2.41 1.90 1.74	2.05 2.03 1.69 1.58	1.58 1.46 1.11 1.01	2.02 1.74 1.30 1.23	106 96.5 91.2 83.0	127 126 126 117	208 193 176 159	230 199 178 167	
Electro Power Pac	373	15761 15761 15762 15762 15762	3,4,7 3,5,7 3,4,7 3,4,6,7 3,5,6,7	114.9 114.9 121.9 230.0 230.0	2.24 2.24 2.24 2.24 2.24	3.55 13.0 4.2 1.45 5.0	3.5 13.3 4.1 1.4 4.8	2.35 9.0 2.8 0.93 3.2	2.2 9.0 2.7 0.9 2.75	9.1 33 10.8 0.37 1.3	8.4 32 9.8 0.33 1.2	5.8 22. 6.9 0.23 0.79	7.0 28 8.53 0.28 0.87	298 1093 397 48.8 168	424 1611 500 67.9 234	628 2406 842 99.6 342	650 2661 898 106 325	
Grimes Co.	Series 550	260 317		134.6 167.6	8.88 8.88	3.5 2.0	3.2 1.9	2.1 1.25	2.0 1.25	8.97 5.13	7.64 4.54	5.19 3.09	6.32 3.95	403 358	533 490	771 711	811 786	
Honeywell	Wingtip AG 113C Light Assembly	7792		124.5	0.0267	8	8	5.6	5.6	20.5	19.1	13.8	17.7	788	1138	1752	1942	
In-Flight Devices	IVI Master Unit	16JE6-3	2	122.6 122.6 138.4	10.24 10.24 10.24	1.2 1.2 1.0	1.13 1.15 0.95	0.774 0.76 0.65	0.703 0.71 0.55	3.07 3.08 2.56	2.70 2.74 2.27	1.91 1.88 1.60	2.22 2.24 1.74	114 115 122	156 158 167	235 232 251	236 238 236	
Whelan	2500	25-1496-1 25-1496-2		135.1 135.1	11.31 11.31	2.7 2.45	2.5 2.3	1.55 1.5	1.45 1.4	6.92 6.28	5.97 5.49	3.83 3.70	4.58 4.42	314 285	419 386	573 554	592 572	
Winn Co.	Omnistrobe	2-2 2-3 2-4 2-5		91.4 91.4 91.4 91.4	10.43 10.43 10.43 10.43	2.1 1.7 1.58 1.55	2.05 1.6 1.46 1.52	1.32 1.01 0.90 0.93	1.25 0.98 0.92 0.92	5.38 4.36 4.05 3.97	4.89 3.82 3.48 3.63	3.26 2.49 2.22 2.30	3.95 3.10 2.91 2.91	111.6 90.46 84.0 82.37	157.2 122.8 111.9 116.7	223.4 170.7 152.1 157.6	233.9 183.6 172.3 172.3	

Notes 1 Special 160 µF Capacitor 2 Viewed at 90° with respect to preceeding measurement 3 with GE FT-91 Flash Tube 4 Jelec = 5.47 Joules 5 Jelec = 17.3 Joules 6 No Reflector 7 Not Designed for Aircraft

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION WASHINGTON, D. C. 20546

OFFICIAL BUSINESS
PENALTY FOR PRIVATE USE \$300

FIRST CLASS MAIL



O2U O01 27 51 3DS 71070 00903 AIR FORCE WEAPONS LABORATORY /WLOL/ KIRTLAND AFB, NEW MEXICO 87117

ATT E. LOU BOWMAN, CHIEF, TECH. LIBRARY

POSTMASTER: If Under

If Undeliverable (Section 15: Postal Manual) Do Not Retu

"The aeronautical and space activities of the United States shall be conducted so as to contribute . . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."

—NATIONAL AERONAUTICS AND SPACE ACT OF 1958

NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

TECHNICAL REPORTS: Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

TECHNICAL NOTES: Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

TECHNICAL MEMORANDUMS:

Information receiving limited distribution because of preliminary data, security classification, or other reasons.

CONTRACTOR REPORTS: Scientific and technical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

TECHNICAL TRANSLATIONS: Information published in a foreign language considered to merit NASA distribution in English.

SPECIAL PUBLICATIONS: Information derived from or of value to NASA activities. Publications include conference proceedings, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

TECHNOLOGY UTILIZATION
PUBLICATIONS: Information on technology
used by NASA that may be of particular
interest in commercial and other non-aerospace
applications. Publications include Tech Briefs,
Technology Utilization Reports and
Technology Surveys.

Details on the availability of these publications may be obtained from:

SCIENTIFIC AND TECHNICAL INFORMATION OFFICE

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
Washington, D.C. 20546